

Compression Field Analysis of Fiber-Reinforced Concrete Based on Cracked Membrane Model

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26 reinforced FRC panels satisfactorily but fails to capture failures governed by sliding of cracks in
27 elements containing high fiber dosages.

28 **Keywords:** fiber reinforced concrete; in-plane shear and normal forces; plane stress; compression
29 field approaches; shear strength; load-deformation analysis.

30 INTRODUCTION

31 It is widely recognized in academia and practice that the addition of steel fibers to concrete may
32 significantly increase the shear strength of reinforced concrete. In certain cases, steel fibers have the
33 capacity to completely replace conventional shear ligature reinforcement¹. This is particularly
34 appealing from an economic and architectural perspective, as the replacement of manually placed
35 steel bar reinforcement with an equivalent dosage of fiber reinforced concrete (FRC) would be
36 extremely beneficial to a project's construction speed, and eliminating the installation of conventional
37 reinforcement would facilitate the construction of free form, non-orthogonal beam and slab
38 geometries, which may otherwise be difficult with conventional reinforcing bar layouts and formwork
39 systems.

40 However, using practical and economical fiber dosages, FRC typically exhibits a softening response
41 in tension after cracking has occurred. Such behavior at the material constitutive level has the
42 potential to lead to a localization of the deformations in single cracks. On the other hand, classical
43 design philosophy in structural concrete relies on members displaying strain hardening behavior in
44 order to achieve a sufficient amount of ductility, which is required to enable the imposed actions to
45 redistribute as the collapse load is approached. This is a prerequisite for the omission of restraint
46 stresses caused by imposed deformations, restrained shrinkage and thermal contraction after
47 hydration etc. in ultimate limit state design. Hence, designers are justifiably hesitant to use FRC
48 without a substantial amount of additional conventional longitudinal steel reinforcement for structural
49 members. Design codes such as AS3600-2018² have imposed severe limits on the use of FRC for
50 such purposes. This reluctance of using fiber only solutions is certainly justified for elements stressed

51 in tension, where the addition of moderate amounts of fibers may even have a detrimental effect on
52 the ductility of girders containing conventional longitudinal reinforcing steels^{3,4}. However, it is less
53 understandable for elements subjected to shear, where conventionally reinforced concrete girders, in
54 compliance with international design codes, usually contain transverse reinforcement ratios in the
55 order of about one quarter of the amount required to achieve a ductile behavior in tension (see
56 References^{5,6}). In fact, many experimental campaigns⁷⁻¹¹ have shown that this is sufficient to avoid
57 premature, brittle shear failures, even if activating the shear resistance requires substantial stress
58 redistribution in the web. Furthermore, several experimental studies on the shear behavior of FRC
59 beams¹²⁻¹⁵ indicate that brittle shear failures can be mitigated and transformed to ductile flexural
60 failures by adding a sufficient dosage of steel fibers to the concrete, without providing additional
61 conventional transverse stirrup reinforcement.

62 Indeed, several recently revised international codes of practice now contain provisions, which permit
63 designers to rely on steel fibers in carrying a portion of the imposed one-way shear in structural
64 concrete^{2,16-18}. However, the conditions for such applications are restrictive, typically limiting the use
65 of steel fibers to resist relatively low shear forces (e.g. 30% of the total shear according to
66 Reference¹⁶) and requiring uneconomically high minimum fiber contents (e.g. $\rho_{f,\min} = 0.75\%$
67 according to ACI-318¹⁸) which may impede the workability of fresh concrete. Such severe restrictions
68 for the use of steel fibers as the only mechanism in carrying shear in structural concrete are pertinent
69 and should be respected, as long as existing models for the global shear strength of fiber reinforced
70 concrete members are semi-empirical. Generally, these design approaches assume that the total shear
71 V_R , is partially resisted by a component taken by the plain concrete, V_{Rc} , a component taken by the
72 fibers, V_{Rf} , and, where applicable, a component resisted by the transverse reinforcement, V_{Rs} , resulting
73 in a total resistance $V_R = V_{Rc} + V_{Rf} + V_{Rs}$. The V_{Rs} component is usually obtained from a truss model
74 and the components V_{Rc} and V_{Rf} are typically calibrated on tests. A first step in the direction of a
75 mechanically based model for the shear strength of FRC was made by the alternative approach
76 mentioned in the explanations to the *fib* Model Code 2010¹⁹⁻²¹, which includes a fiber contribution

77 V_{Rf} that essentially corresponds to a compression field approach. However, the inclination of the
78 compression field and the crack width required in determining the effective fiber bridging stresses
79 are evaluated using semi-empirical equations. Furthermore, the empirical concrete contribution, V_{Rc} ,
80 which is somewhat related to the shear at the commencement of diagonal cracking, is still added to
81 obtain the global shear strength. This is in contrast to well-established approaches for conventionally
82 reinforced structural concrete, where if the minimum shear reinforcement discussed above is
83 provided, a safe and economical design is possible using a simple variable angle truss model
84 approach, based on the lower bound theorem of the theory of plasticity^{19,22,23}, without the need for
85 empirical concrete contributions. In order to fully exploit the potential of using steel fibers as shear
86 reinforcement, mechanically consistent models for the design and analysis of FRC in shear must be
87 developed.

88 **RESEARCH SIGNIFICANCE**

89 It is well established that the addition of fibers to concrete increases the shear strength of a reinforced
90 concrete member. Available models describing the behavior of fiber reinforced concrete (FRC) in
91 shear are, however, founded on semi-empirical rationale. This paper presents a mechanically
92 consistent rotating crack model to describe the behavior of FRC elements in plane stress. The model
93 is based on the Cracked Membrane Model^{23,24}, a compression field approach for conventionally
94 reinforced concrete, extended herein to account for fiber reinforcement. Through this study, improved
95 methods of analysis of FRC members subjected to shear can be rationally achieved.

96 **COMPRESSION FIELD APPROACHES**

97 **Compression field approaches for conventionally reinforced concrete elements**

98 The load-deformation behavior of diagonally cracked reinforced concrete elements can be readily
99 analyzed using compression-field approaches. These approaches were developed several decades
100 ago^{25,26}, and systematically satisfy equilibrium conditions and compatibility requirements. The basic
101 concepts of such models are illustrated in Fig. 1 for the special case of orthogonal reinforcement,

102 whereby the horizontal and vertical reinforcement are assumed to coincide with the x - and z -
 103 directions, respectively.

104 Consider the element shown in Fig. 1a. Noting that the sum of forces in the concrete and
 105 reinforcement must correspond to the applied loads (Fig. 1d), the following two equivalent sets of
 106 equilibrium conditions are obtained

$$\begin{aligned}
 \sigma_x &= \sigma_{cx} + \sigma_{sx} = \sigma_{c3} \cos^2 \theta + \sigma_{c1} \sin^2 \theta + \rho_x \sigma_{sx} & \rho_x \sigma_{sx} &= \sigma_x + \tau_{zx} \cot \theta - \sigma_{c1} \\
 \sigma_z &= \sigma_{cz} + \sigma_{sz} = \sigma_{c3} \sin^2 \theta + \sigma_{c1} \cos^2 \theta + \rho_z \sigma_{sz} & \text{or} & \rho_z \sigma_{sz} &= \sigma_z + \tau_{zx} \tan \theta - \sigma_{c1} \\
 \tau_{zx} &= \tau_{czx} + \tau_{szx} = (\sigma_{c1} - \sigma_{c3}) \sin \theta \cos \theta & & \sigma_{c3} &= -\tau_{zx} (\tan \theta + \cot \theta) + \sigma_{c1}
 \end{aligned} \tag{1}$$

108 if the reinforcing bars are assumed to only carry uniaxial stresses along their longitudinal axis, i.e.
 109 neglecting dowel action such that $\tau_{szx} = 0$. In Fig. 1d, ρ_x and ρ_z are the geometric reinforcement
 110 ratios in the x - and z -directions, respectively, σ_c and σ_s are the stresses in the concrete and
 111 reinforcement, respectively and θ is the principal compressive stress and strain direction.
 112 Compatibility requires that the strains (Fig. 1b) are related through

$$\cot^2 \theta = \frac{\varepsilon_z - \varepsilon_3}{\varepsilon_x - \varepsilon_3} = \frac{\varepsilon_1 - \varepsilon_x}{\varepsilon_x - \varepsilon_3} = \frac{\varepsilon_z - \varepsilon_3}{\varepsilon_1 - \varepsilon_z} \tag{2}$$

114 i.e., the state of strain is fully determined by three independent unknowns. Expressing the concrete
 115 stresses (σ_{c1}, σ_{c3}) and the stresses in the reinforcement (σ_{sx}, σ_{sz}) in terms of any 3 non-collinear
 116 strains, e.g. ($\varepsilon_x, \varepsilon_z, \varepsilon_3$), and using constitutive relationships relating stresses and strains of concrete
 117 and reinforcement, respectively, the states of stress and strain can be determined iteratively by solving
 118 the 3 equations in Eqs. (1) for the 3 unknown strains.

119 The Equations presented in Eqs. (1) may be simplified by neglecting the tensile stresses in the
 120 concrete (i.e. $\sigma_{c1} = 0$, refer to Fig. 1c), as typically assumed in design²³. For linear elastic behavior
 121 for the reinforcement and concrete, with $n = E_s/E_c$, Baumann's closed-form equation for the
 122 principal compressive stress and strain direction is obtained²⁷:

$$\tan^2 \theta \rho_x (1 + n \rho_z) + \tan \theta \rho_x \frac{\sigma_z}{\tau_{xz}} = \cot^2 \theta \rho_z (1 + n \rho_x) + \cot \theta \rho_z \frac{\sigma_x}{\tau_{xz}} \tag{3}$$

124 Having determined the principal compressive stress and strain direction θ from Eq. (3), obtaining the
125 response (strains for any given applied stresses) by inserting θ in Eqs. (1) is straightforward.

126 Neglecting the tensile strength of the concrete, f_{ct} , in any direction corresponds to the assumption of
127 fictitious, rotating, stress-free cracks which open orthogonally, and hence, the principal compressive
128 directions of average strains θ_ϵ and concrete stresses θ_σ coincide with the direction of the fictitious
129 cracks θ_r ($\theta_r = \theta_\epsilon = \theta_\sigma = \theta$), refer to Fig. 1. Predictions obtained using this type of classical compression
130 field approach – even if using nonlinear constitutive relationships for concrete and reinforcement²⁸ –
131 tend to overestimate the observed deformations since rotating cracks are considered and tension
132 stiffening effects are neglected. Furthermore, they typically overestimate the ultimate load unless
133 compression softening of the concrete is accounted for²³.

134 Vecchio and Collins proposed the Modified Compression Field Theory (MCFT)²⁹ to overcome these
135 weaknesses. In the MCFT, tension stiffening is implicitly taken into account by formulating
136 equilibrium in terms of average stresses between the cracks, including average tensile stresses in the
137 concrete (Fig. 1d). However, by treating the assumed average tensile stresses in the concrete as a
138 material property, depending only on the principal tensile strain, the respective empirical constitutive
139 equations disregard the fact that tension stiffening strongly depends on the rebar diameter and the
140 steel reinforcement ratios³⁰. Consequently, such an approach is unable to yield direct information on
141 the maximum stresses at the cracks nor on crack spacings or crack widths. The same remark applies
142 to other models formulating equilibrium in terms of average stresses between cracks^{31,32}.

143 Extending this methodology for FRC is not entirely appropriate as the most important property when
144 considering the design of a structural member manufactured with FRC is its post cracking, or residual,
145 tensile strength. The strength that is offered by the fibers across a crack is dependent on the width of
146 the crack. Hence for a reliable analysis of FRC elements, information on the crack opening must be
147 established. This is why compression field models formulating equilibrium in terms of stresses at the
148 cracks, which can yield direct information on crack spacing and openings, are much better suited for
149 the analysis of FRC elements in shear. One such model is the Cracked Membrane Model (CMM)^{23,24}.

150 In its general formulation (CMM-F), this model allows treating cracks as fixed and interlocked rather
151 than as rotating and stress-free while accounting for shear and normal stresses on the crack faces as
152 well as bond stresses transferred between the concrete and steel reinforcement. Equilibrium
153 conditions are expressed in terms of stresses at the cracks (see Fig. 2), and crack spacings and tensile
154 stresses between the cracks are determined from basic mechanical principles adopted from the
155 Tension Chord Model (TCM)⁵. However, calculations using this general model are tedious, and the
156 assumption of interlocked cracks with a fixed direction is of limited use for design purposes, since
157 this is equivalent to allowing substantial tensile stresses to develop in the concrete between the cracks.
158 Therefore, the simplified version of the model considering rotating, stress-free cracks, the CMM-R,
159 is considered in the following^{23,24}. This model has been successfully implemented in FE-codes by
160 several researchers³³⁻³⁵, including a generalization to the analysis of shell elements using a sandwich
161 model approach³⁶⁻³⁸.

162 In the CMM-R, contrary to the CMM-F, cracks are assumed to be stress free and open orthogonally.
163 These two assumptions greatly simplify the calculations without losing the direct information on the
164 local stresses at the cracks and crack openings, since equilibrium conditions are expressed in terms
165 of stresses at the cracks. In particular, Eqs. (1) can be simplified neglecting the tensile stresses in the
166 concrete at the cracks (i.e. setting $\sigma_{c1} = 0$, refer to Fig. 1c) and remain valid if stresses in the concrete
167 and reinforcement at the cracks, as well as $\theta_r = \theta$ are inserted, and a closed form solution for the
168 diagonal crack spacing, s_r , can be established based on the theoretical maximum crack spacings for
169 uniaxial tension in the reinforcement directions, s_{rx0} and s_{rz0} , respectively^{23,24}. The latter are obtained
170 observing that while $\sigma_{c1} = 0$ at the cracks, tensile stresses are transferred to the concrete between the
171 cracks by bond shear stresses, and that the tensile stresses in the concrete cannot exceed f_{ct} midway
172 between two cracks. Using the TCM⁵, which assumes constant rigid-plastic bond stresses (i.e.
173 independent of the slip) prior to yielding of the reinforcement, the maximum theoretical crack
174 spacings for uniaxial tension are given by

$$s_{rx0} = \frac{\emptyset_x f_{ct} (1 - \rho_x)}{2\tau_{b0}\rho_x} \approx \frac{\emptyset_x f_{ct}}{2\tau_{b0}\rho_x} = \frac{\emptyset_x}{4\rho_x}, \quad s_{rz0} = \frac{\emptyset_z f_{ct} (1 - \rho_z)}{2\tau_{b0}\rho_z} \approx \frac{\emptyset_z}{4\rho_z} \quad (\tau_{b0} = 2f_{ct}) \quad (4)$$

176 where \emptyset_x and \emptyset_z are the bar diameters in the x - and z -directions, respectively, $\tau_{b0} = 2f_{ct} = 0.6f_c^{2/3}$
177 is the bond shear stress for elastic reinforcement according to the TCM, with f_c being the concrete
178 cylinder strength in MPa units. Note that the minimum theoretical crack spacing is half the
179 corresponding maximum crack spacing, since at the limit, a new crack may form when $\sigma_{c1} = f_{ct}$ is
180 attained midway between two cracks. This uncertainty cannot be avoided, even with more realistic,
181 complex bond shear stress-slip relationships, which justifies the use of a simplified, yet mechanically
182 consistent model like the TCM⁵. In order to determine the crack spacing in biaxially reinforced
183 elements, the stresses in the concrete between the cracks can be determined by superimposing the
184 concrete stresses at the crack with the stresses transferred to the concrete between the cracks by bond
185 shear stresses from both sets of orthogonally placed reinforcing bars. Noting that the crack spacings
186 are geometrically related through $s_r = s_{rx} \sin \theta = s_{rz} \cos \theta$ (see Fig. 2a), and that just as for uniaxial
187 tension, the maximum diagonal crack spacing s_{r0} is attained when $\sigma_{c1} = f_{ct}$ occurs at the center
188 between two cracks, s_{r0} can be determined analytically^{23,24}. A good upper bound approximation for
189 s_{r0} is given by

$$s_{r0} = \frac{1}{\frac{\sin \theta}{s_{rx0}} + \frac{\cos \theta}{s_{rz0}}} \quad (5)$$

191 Eq. (5) is the same equation postulated by Vecchio and Collins²⁹. Like in uniaxial tension^{5,6}, the
192 diagonal crack spacing may also vary by a factor of two, i.e. $s_r = \lambda s_{r0}$, with $\lambda = 0.5 \dots 1.0$ ^{23,24}. The
193 crack opening, w_r , can be determined from the principal tensile strain, ϵ_1 , and the diagonal crack
194 spacing $s_r = \lambda s_{r0}$, i.e.

$$w_r = s_r (\epsilon_1 - \epsilon_{c1}) \approx s_r \epsilon_1 = \lambda s_{r0} \epsilon_1 \quad (6)$$

196 Eq. (6) is simplified by neglecting the tensile strains within the concrete between the cracks, ε_{c1} ,
197 without significantly affecting accuracy, since typically, $\varepsilon_{c1} \ll \varepsilon_1$.
198 Knowing the crack spacing (or rather, selecting a value between its theoretical upper and lower
199 boundaries), the CMM-R analysis can be carried out as outlined above for classic compression field
200 models, i.e. solving the Eqs. (1) (with $\sigma_{c1} = 0$) to obtain the state of strain (3 unknowns) for any given
201 set of applied stresses. However, constitutive relationships for the concrete and the reinforcement
202 expressing the (maximum) stresses at the cracks in terms of average strains are required. The TCM
203 allows for determining these relationships consistently as a function of the crack spacing⁵, with
204 closed-form analytical solutions e.g. when using a bilinear stress-strain relationship for the bare bar
205 reinforcement. For linear elastic reinforcement, a closed form solution for the crack inclination θ_r ,
206 similar to Eq. (3) but accounting for tension stiffening in a consistent manner, can be derived, and is
207 a useful tool for verifications of the serviceability limit state, particularly for crack widths³⁹. For more
208 details, the reader is referred to References^{23,24}.

209 **Compression field approach for FRC elements**

210 Adapting the CMM-R to account for fiber reinforcement (CMM-Rf) is straightforward, since
211 equilibrium is formulated at the cracks and fiber stresses at cracks can be determined from the crack
212 kinematics that are directly obtained from the model. Hence, equilibrium can simply be expressed
213 using Eqs. (1) (Fig. 1d), and setting $\sigma_{c1} = \sigma_{cf}$, where σ_{cf} is the effective (residual) fiber stresses at
214 the cracks. These stresses act perpendicularly to the crack face and hence, in the direction of the
215 principal concrete tensile stresses. The residual stresses provided by the fibers are dependent on the
216 crack opening, i.e. $\sigma_{c1} = \sigma_{cf}(u)$, where $u = w_r$ and can be obtained from Eq. (6).

217 However, the crack spacings (refer to Eq. (4)) need to be adapted to account for the fiber bridging
218 stresses. As outlined above for conventional reinforced concrete, moving away from a crack along a
219 reinforcing bar, the tensile stresses in the surrounding concrete will gradually build up, as a result of
220 bond shear stresses transferred from the reinforcement, to a value up to f_{ct} . This distance is shorter in

221 FRC than in concrete without fibers since in addition to bond, the fiber stresses $\sigma_{c1} = \sigma_{cf}$ at the crack
 222 are transmitted to the concrete over the fiber embedment length $l_{bf} \leq l_f/2$. Using the same
 223 assumptions as in the TCM, the following crack spacings for uniaxial tension in the reinforcement
 224 directions, accounting for additional fiber reinforcement, result³:

$$225 \quad s_{rxf0} = \frac{\emptyset_x}{4\rho_x} \left(1 - \frac{\sigma_{cf}}{f_{ct}} \right), \quad s_{rzf0} = \frac{\emptyset_z}{4\rho_z} \left(1 - \frac{\sigma_{cf}}{f_{ct}} \right) \quad (\tau_{b0} = 2f_{ct}) \quad (7)$$

226 Using Eqs. (7), the maximum diagonal crack spacing can be determined as for conventionally
 227 reinforced panels, again with $s_{rf} = \lambda s_{rf0}$ ($\lambda = 0.5 \dots 1.0$)^{23,24}. The diagonal crack spacing
 228 $s_{rf} = s_{rxf} \sin \theta = s_{rzf} \cos \theta$ accounting for biaxial reinforcement and fibers, can be determined based on
 229 the theoretical maximum diagonal crack spacing, s_{rf0} , as follows:

$$230 \quad s_{rf0} = \frac{1}{\frac{\sin \theta}{s_{rxf0}} + \frac{\cos \theta}{s_{rzf0}}} = \frac{1 - \sigma_{cf}/f_{ct}}{\frac{\sin \theta}{s_{rx0}} + \frac{\cos \theta}{s_{rz0}}}, \quad s_{rf} = (0.5 \dots 1.0) s_{rf0} \geq l_f \quad (8)$$

231 where θ is the principal compressive stress direction at the onset of initial cracking.
 232 Note that for high fiber dosages, Eqs. (7) can yield very small or even negative crack spacings (for
 233 $\sigma_{cf} > f_{ct}$); however, this is physically impossible since for small crack spacings, these high fiber
 234 stresses cannot be activated. By this, an imposed lower bound limit for the crack spacing is taken as
 235 the length of the fiber (i.e. $s_{rf} \geq l_f$) as indicated in Eq. (8). Since the crack opening is typically small
 236 during the crack formation process, a reasonable approximation of the crack spacing is obtained by
 237 setting $\sigma_{cf} \approx \sigma_{cf0}$ in Eqs. (7) and (8), where σ_{cf0} denotes the fiber effectiveness, corresponding to
 238 the maximum value of residual fiber stresses at small crack openings as further outlined below (refer
 239 to Eq. (11)).

240 For members without conventional reinforcement in either direction, Eq. (8) cannot be used. Rather,
 241 for members with high fiber dosages, the crack spacing will be in the order of $s_{rf} \approx l_f$, and for low
 242 fiber dosages, localized cracking is expected to occur unless the crack spacing in either direction is

243 governed by adjacent elements, such as the chords in the case of the web of a girder. In the latter case,
 244 the diagonal crack spacing in the web can be estimated geometrically, using the crack spacing $s_{rx,TC}$
 245 in the tension chord:

$$246 \quad s_{rf} = s_{rx,TC} \sin \theta \quad (9)$$

247 Knowing the crack spacing, the value of σ_{cf} can be determined for a given state of strain (and hence
 248 crack opening) using Eq. (6) and any suitable FRC material constitutive stress vs crack opening
 249 relationship. In this paper, the simple mechanical model proposed by Pfy1^{3,40}, illustrated in Fig. 3, is
 250 adopted to estimate the residual characteristics of the FRC for the sake of simplicity. It is
 251 acknowledged that more accurate expressions for the residual capacity of FRC may be obtained using
 252 fiber stress vs crack opening characteristics derived directly from standard material tests on the actual
 253 FRC used⁴¹, or through the use of more refined/sophisticated constitutive models that are available in
 254 the literature^{42,43}. Also, better predictions are likely^{42,43} to be obtained using statistical models to predict
 255 material parameters; however, this is beyond the scope of this paper.

256 While Pfy1's model considers fiber activation followed by a pull-out phase, only the latter is
 257 considered in the following for simplicity (Fig. 3b). Hence, fiber stresses amount to

$$258 \quad \frac{\sigma_{cf}}{\sigma_{cf0}} = \left(1 - \frac{2u}{l_f} \right)^2 \quad \left(u \leq \frac{l_f}{2} \right) \quad (10)$$

259 In the derivation of Eq. (10), fiber snubbing effects⁴³ were neglected. The term σ_{cf0} denotes the *fiber*
 260 *effectiveness* of the FRC, which can be expressed as

$$261 \quad \sigma_{cf0} = K_f \frac{\rho_f \tau_{bf} l_f}{d_f} \quad (11)$$

262 if the pullout resistance of the individual fibers is assumed to be independent of their inclination θ_f
 263 with respect to the outer normal to the crack surface. This assumption can be justified if fibers oriented
 264 close to the crack plane (e.g. $\theta_f > \pi/3$ as proposed by Foster⁴⁴) are considered ineffective, as
 265 assumed in the following. In Eq. (11), ρ_f is the total volumetric fiber content, K_f is a fiber

266 orientation factor explained in the Appendix, and τ_{bf} is the average bond stress between the fibers
267 and surrounding matrix along the shorter embedded length, which accounts for stress transfer between
268 the fibers and the matrix by anchorage provided by end hooks, taken as $\tau_{bf} = 2f_{ct} = 0.6f_c^{2/3}$ (in MPa
269 units).

270 In the discussion above a fundamental assumption is made on the failure mechanism of FRC in that
271 the fibers pull-out of the matrix. If the axial stresses developed within the fibers exceeded the ultimate
272 tensile strength of the fibers, f_{tf} , the fibers would fracture before being pulled out. Pfy's model
273 above can be extended for the case of fiber fracture, noting that fibers longer than

$$274 \quad l_{fbu} = \frac{f_{tf} d_f}{4\tau_{bf}} \quad (12)$$

275 would fracture before completely pulling out. However, the fiber geometry (l_f, d_f) and properties
276 are usually tuned to avoid this behavior. Therefore, fiber fracture is not considered in the development
277 of the model.

278 Furthermore, it has been observed in tests on FRC members where single dominant/localized cracks
279 form that the number of fibers bridging failure surfaces are generally lower than the total expected
280 number of fibers⁴¹. It is thought that the crack path follows the easiest propagation route through the
281 FRC matrix and often propagates through the inherent flaws of the concrete (such as pores) and near
282 the end or around individual fibers. In shear critical FRC beams, Amin and Foster⁴⁵ included a fiber
283 dispersion factor of 0.82 which is applied to the average residual tension that is offered by the fibers
284 to account for these observations. For FRC members containing steel rebar where many cracks are
285 expected to form, it may be argued that the initial cracks may contain, on average, fewer fibers than
286 theoretically expected. However, after first cracking, the location of the subsequent cracks depends
287 more-so on the development of bond between the FRC matrix and steel rebar than the presence of the
288 fibers. Assuming many cracks smeared along the element, and large crack surfaces, the influence of
289 those few initial cracks with statistically fewer fibers becomes negligible to the overall response of

290 the element. Note that it has also been observed in tension tie tests that one of these initial cracks is
 291 more likely to develop into the failure surface of the member, however these failure surfaces tend to
 292 localize well after peak load has been attained^{46,47}.

293 The CMM-Rf analysis including fibers can be carried out just as outlined above for the CMM-R
 294 without fibers, but solving Eq. (1) with $\sigma_{c1} = \sigma_{cf}(u)$ instead with $\sigma_{c1} = 0$ to obtain the state of strain
 295 (3 unknowns) for a given set of applied stresses, where the crack opening $u = w_r$ can be determined
 296 from Eq. (6), using the crack spacing given by Eq. (8).

297 **Simplified compression field approach for FRC webs**

298 For elements without vertical reinforcement nor vertically applied stresses, as is the case for the webs
 299 of girders without conventional shear stirrup reinforcement, the compression field analysis presented
 300 above can be simplified. Using Eqs. (1) with $\sigma_{c1} = \sigma_{cf}$ and setting $\rho_z = \sigma_z = 0$, one obtains (Fig. 4c)

$$\rho_x \sigma_{sx} - \sigma_x = \tau_{zx} \cot \theta - \sigma_{cf} = \tau_{zx} \left(\frac{\tau_{zx}}{\sigma_{cf}} - \frac{\sigma_{cf}}{\tau_{zx}} \right) = \frac{\tau_{zx}^2}{\sigma_{cf}} - \sigma_{cf}$$

$$301 \quad \tan \theta = \frac{\sigma_{cf}}{\tau_{zx}} \quad (13)$$

$$\sigma_{c3} = -\frac{\tau_{zx}^2}{\sigma_{cf}}$$

302 Knowing σ_{cf} , the state of stress at the cracks for any applied stresses (σ_x, τ_{zx}) is thus uniquely
 303 determined by equilibrium alone. However, σ_{cf} is a function of the crack opening, which in turn is
 304 dependent on the principal tensile strain (see Eq. (6)). Rather than using the general solution
 305 procedure, a more direct way consists of: (i) assuming a value of σ_{cf} ; (ii) determining the
 306 corresponding values of ε_x , ε_3 and θ from Eqs. (13), using the constitutive relationships of
 307 concrete and longitudinal reinforcement; (iii) determining the principal tensile strain using the
 308 compatibility condition Eq. (2), i.e. $\varepsilon_1 = \varepsilon_x + (\varepsilon_x - \varepsilon_3) \cot^2 \theta$; (iv) calculating the resulting crack
 309 opening using Eq. (6); (v) determining the fiber stress corresponding to the obtained crack opening

310 from Eq. (10); and (vi) iterating through steps (i)-(v) until the resulting fiber effective stress converges
 311 with the initially assumed value.

312 Assuming linear elastic behavior of the concrete in compression and, where applicable, for the
 313 longitudinal reinforcement in tension, further simplifications are possible. This is outlined below for
 314 two cases of practical interest, representing possible situations in the webs of girders.

315 Consider first a uniaxially reinforced FRC element loaded in pure shear, i.e. $\rho_z = 0$ and $\sigma_x = \sigma_z = 0$.

316 Using the equilibrium conditions, Eqs. (13), one attains

$$317 \quad \varepsilon_3 = \varepsilon_{c3} - \frac{n\tau_{zx}^2}{\sigma_{cf}E_s}, \quad \varepsilon_x = \frac{\tau_{zx}}{\rho_x E_s} \left(\frac{\tau_{zx}}{\sigma_{cf}} - \frac{\sigma_{cf}}{\tau_{zx}} \right) \quad (14)$$

318 and inserting Eqs. (14) into the compatibility condition given by Eq. (2), i.e. $\varepsilon_1 = \varepsilon_x + (\varepsilon_x - \varepsilon_3) \cot^2 \theta$,

319 the principal tensile strain amounts to

$$320 \quad \varepsilon_1 = \frac{\tau_{zx}^2}{\rho_x E_s \sigma_{cf}} \left(\frac{\tau_{zx}^2}{\sigma_{cf}^2} (1 + n\rho_x) - \frac{\sigma_{cf}^2}{\tau_{zx}^2} \right) \quad (15)$$

321 Hence, the crack opening $w_r = s_r \varepsilon_1$ (see Eq. (6)) can be expressed in a closed form, and the states of
 322 stress and strain are found by simply solving the equation

$$323 \quad \sigma_{cf} \left(s_{rf} \cdot \frac{\tau_{zx}^2}{\rho_x E_s \sigma_{cfi}} \left(\frac{\tau_{zx}^2}{\sigma_{cfi}^2} (1 + n\rho_x) - \frac{\sigma_{cfi}^2}{\tau_{zx}^2} \right) \right) = \sigma_{cfi} \quad (16)$$

324 for an initially assumed value σ_{cfi} of the fiber stress, using $\sigma_{cf}(u)$ defined by Eq. (10) (or any other,
 325 more refined FRC constitutive model^{42,43}).

326 Consider next an element with $\rho_x = \rho_z = 0$ of the web of a girder with stiff chords, whose strains are
 327 typically dominated by bending moments, but generally also depend on the applied shear stresses,
 328 $\varepsilon_x = g(\tau_{zx})$. The equilibrium conditions, Eqs. (13), in this case yield the strains

$$329 \quad \varepsilon_3 = \varepsilon_{c3} - \frac{n \cdot \tau_{zx}^2}{\sigma_{cf} E_s}, \quad \varepsilon_x = g(\tau_{zx}) \quad (17)$$

330 and inserting Eqs. (17) into the compatibility condition Eq. (2), i.e. $\varepsilon_1 = \varepsilon_x + (\varepsilon_x - \varepsilon_3) \cot^2 \theta$, the
 331 principal tensile strain amounts to

$$\varepsilon_1 = g(\tau_{zx}) \left(1 + \frac{\tau_{zx}^2}{\sigma_{cf}^2} \right) + \frac{n\tau_{zx}^4}{E_s \sigma_{cf}^3} \quad (18)$$

The crack opening $w_r = s_r \varepsilon_1$ (see Eq. (6)) can again be expressed in closed form, and the states of stress and strain are determined by solving the equation

$$\sigma_{cf} \left(s_{rf} \cdot \left(g(\tau_{zx}) \left(1 + \frac{\tau_{zx}^2}{\sigma_{cfi}^2} \right) + \frac{n\tau_{zx}^4}{E_s \sigma_{cfi}^3} \right) \right) = \sigma_{cfi} \quad (19)$$

for an initially assumed value σ_{cfi} , as above.

DISCUSSION

Figures 5 and 6 present the results of sample compression field analyses for the two special cases discussed above, i.e. a uniaxially reinforced element (Fig. 5) and a FRC element without any conventional reinforcement, but subjected to imposed horizontal strains (Fig. 6), respectively. Two sets of calculations, both considering elastic behavior of the reinforcement, were carried out for each case, i.e. Figs. 5 and 6, considering (i) linear elastic behavior of the concrete, i.e. simply solving Eqs. (16) and (19), respectively, with $n = E_s / E_c = 6$ (dashed lines), and (ii) following the simplified approach for FRC webs outlined in the previous section, using the following parabolic stress-strain relationship accounting for compression softening for the concrete (solid lines)^{23,24}:

$$\sigma_{c3} = - \left(\frac{2\varepsilon_{c3}}{\varepsilon_{c0}} - \left(\frac{\varepsilon_{c3}}{\varepsilon_{c0}} \right)^2 \right) \frac{f_c^{2/3}}{0.4 + 30\varepsilon_1} \quad \text{i.e.} \quad \varepsilon_{c3} = \varepsilon_{c0} \left(1 - \sqrt{1 + \frac{\sigma_{c3}}{f_c^{2/3}} (0.4 + 30\varepsilon_1)} \right) \quad (20)$$

In Eq. (20) f_c is the concrete cylinder strength in MPa and ε_{c0} is the strain corresponding to the peak compressive stress $f_{cred} = f_c^{2/3} / (0.4 + 30\varepsilon_1)$, with $\varepsilon_{c3} \geq \varepsilon_{c0} \approx -0.002$. The behavior is analyzed for a fiber effectiveness equivalent to $\sigma_{cf0} = 2.0$ MPa (0.29 ksi), for both the maximum, as well as the minimum theoretically predicted crack spacing, i.e. $\lambda = 1.0$ and $\lambda = 0.5$, respectively, to study the influence on the response. Furthermore, it is assumed that the fiber geometry and matrix strength are tuned such that ruptures of the fibers do not occur. The considered fiber effectiveness translates to a

353 steel fiber dosage of approximately $40 \dots 60 \text{ kg/m}^3$ ($2.5 \dots 3.8 \text{ lb/ft}^3$) ($\rho_f = 0.50 - 0.75\%$ by volume)
 354 depending on the fiber orientation and the specific properties of the matrix and fibers).
 355 The response of the uniaxially reinforced FRC element in pure shear, $\sigma_x = 0$, is presented in Fig. 5,
 356 for a longitudinal reinforcement ratio of $\rho_x = 1.5\%$. The material parameters used in the analyses are
 357 indicated in the Figure. It can be seen that as predicted by the second equation in Eq. (13), the
 358 principal compressive stress direction rotates towards the horizontal x -axis with increasing shear
 359 stresses τ_{zx} , and that $\cot \theta$ increases almost proportionally with τ_{zx} as long as the decrease in fiber
 360 stresses $\sigma_{c1} = \sigma_{cf}$ due to crack opening u is moderate. In the case of *linear elastic* concrete behavior
 361 (dashed lines), maximum loads of $\tau_{zXu} / \sigma_{cf0} = 1.63$ and $\tau_{zXu} / \sigma_{cf0} = 1.93$ are predicted for $\lambda = 1.0$ and
 362 $\lambda = 0.5$, respectively, with strains in the longitudinal reinforcement reaching $1.77 \cdot 10^{-3}$ and
 363 $2.81 \cdot 10^{-3}$. Hence, in this case, depending on the yield strength of the reinforcement, yielding of the
 364 reinforcement would govern failure, as indicated by the intersections of the vertical dotted line and
 365 model resultants shown in Figs. 5a and 5b for a yield strength of 500 MPa (72.5 ksi). Note that while
 366 in an orthogonally reinforced panel, a load increase is possible after yielding of the reinforcement in
 367 one direction (accompanied by higher steel stresses in the other reinforcement direction), such a
 368 redistribution is not possible in a uniaxially reinforced FRC panel due to the strain-softening
 369 characteristics of the FRC. If the yield strength of the reinforcement is sufficiently high, fiber
 370 softening would be the governing mode of failure: as the tensile strains, $\varepsilon_1 = \varepsilon_x + (\varepsilon_x - \varepsilon_3) \cot^2 \theta$,
 371 increase essentially in proportion with $\cot^2 \theta$ and hence, with τ_{zx}^2 , see the second equation in Eq. (13)
 372 , these increasing strains lead to progressively larger crack openings and consequently the ability of
 373 the fibers to transmit tension across cracks is reduced, which in turn requires flatter inclinations until
 374 the point where equilibrium cannot be satisfied. For *nonlinear* concrete behavior (solid lines),
 375 maximum loads of $\tau_{zXu} / \sigma_{cf0} = 1.58$ and $\tau_{zXu} / \sigma_{cf0} = 1.82$ are predicted for $\lambda = 1.0$ and $\lambda = 0.5$,
 376 respectively, with corresponding strains at the peak load in the longitudinal reinforcement reaching

377 $1.56 \cdot 10^{-3}$ and $2.19 \cdot 10^{-3}$. Hence, in this case, reinforcement yielding will not govern unless low
378 strength reinforcement is used and failure occurs when equilibrium cannot be satisfied due to the
379 combined effect of tensile softening of the fiber stresses, as outlined for the linear case, and
380 compression softening of the concrete. The latter is very pronounced, since the compressive strength
381 decreases hyperbolically with increasing tensile strains, ϵ_1 , see Eq. (20) that in turn grow essentially
382 in proportion with $\cot^2 \theta$ and τ_{zx}^2 , see above, just like the concrete compressive stresses, see the third
383 equation in Eq. (13).

384 Noting that cracking under pure shear is expected to occur at $\tau_{zxr} = f_{ct} \approx 0.3 f_c^{2/3} = 3.5 \text{ MPa}$ (0.51 ksi),
385 i.e. $\tau_{zxr} / \sigma_{cf0} \approx 1.75$ (or lower if restraint stresses due to shrinkage and hydration heat are considered),
386 it is seen that in spite of the moderate fiber dosage used, with $\sigma_{cf0} < f_{ct}$, the applied shear stresses τ_{zx}
387 can increase after cracking if the crack spacing is small (near the minimum theoretical crack spacing)
388 with $\tau_{zxu} / \tau_{zxr} \approx (1.82 \dots 1.93) / 1.75 = 1.04 \dots 1.10$, but a brittle failure at cracking would occur for the
389 maximum theoretical crack spacing, with $\tau_{zxu} / \tau_{zxr} \approx (1.58 \dots 1.63) / 1.75 = 0.90 \dots 0.93$.

390 By examining the differences between $\lambda = 1.0$ and $\lambda = 0.5$ in Fig. 5, it is evident that the crack
391 spacing has a pronounced effect on the overall behavior, with differences in predicted ultimate loads
392 ranging between 15...20%, even if the specimen is precracked, or if the cracking load is lower than
393 estimated above. Hence, the inherent uncertainty in the spacing of cracks must be acknowledged
394 when analyzing and designing FRC elements subjected to shear. However, even with larger crack
395 spacings, a strain hardening behavior could be obtained using a higher conventional reinforcement
396 ratio, as in the experiments discussed in the following section.

397 Figure 6 presents the results of a similar analysis for the FRC element without any conventional
398 reinforcement, but subjected to imposed horizontal strains. The material parameters used in the
399 analysis are indicated in the Figure. Horizontal strains were chosen to represent a typical prestressed
400 concrete girder, with an initial prestress of -5 MPa (0.72 ksi) (corresponding to $\epsilon_x = -0.17 \cdot 10^{-3}$) and

401 longitudinal strains increasing only slightly until decompression of the tension chord ($\varepsilon_{x,TC} = 0$),
 402 which is assumed to occur at $\tau_{zx} = 1.5 \text{ MPa}$ (0.22 ksi). After decompression, longitudinal strains
 403 grow faster until the tension chord yields at $\varepsilon_{x,TC} = 2.0 \cdot 10^{-3}$ (e.g. reinforcement with $f_y = 500 \text{ MPa}$
 404 (72.5 ksi) and moderate tension stiffening effect), assumed in the example to occur at $\tau_{zx} = 4 \text{ MPa}$
 405 (0.58 ksi). As usual in design approaches for conventionally reinforced girders, longitudinal strains
 406 at the mid-depth of the web are used as a reference point to predict the overall response¹⁹, i.e.,
 407 neglecting favorable strains of the compression chord, half the longitudinal strains of the tension
 408 chord are imposed to the web, $\varepsilon_x = \varepsilon_{x,TC} / 2$. Assuming a maximum spacing of the cracks in the
 409 tension chord triggering cracks in the web of $s_{r,TC} = 400 \text{ mm}$ (15.75 in.) and a principal compressive
 410 stress direction at cracking of $\theta = 30^\circ$ (a value lower than 45° due to the effect of prestressing), a
 411 maximum diagonal crack spacing of $s_{r,f0} = 200 \text{ mm}$ (7.87 in.) is obtained from Eq. (9). Since strains
 412 are small before cracking of the concrete, which is expected to occur at $\varepsilon_1 = \varepsilon_{ct} = f_{ct} / E_c \approx 10^{-4}$, the
 413 cracking load τ_{zxr} is estimated as the shear stress τ_{zxr} acting at $\varepsilon_x = \varepsilon_{ct}$, corresponding to
 414 $\tau_{zxr} / \sigma_{cf0} \approx 0.80$ here. Note that in a practical case, all the values assumed above would be determined
 415 based on the actual applied prestressing (stress state in web before cracking), the moment-shear
 416 interaction at the critical cross-section M/V , and the tension chord reinforcement layout, using a
 417 sectional analysis.

418 The resulting behavior of the web is generally similar as in the uniaxially reinforced case. Failure is
 419 caused by combined tensile softening of the fibers and compression softening of the concrete in all
 420 cases. For linear elastic concrete, maximum loads amount to $\tau_{zxr} / \sigma_{cf0} = 2.34$ and $\tau_{zxr} / \sigma_{cf0} = 2.85$
 421 for $\lambda = 1.0$ and $\lambda = 0.5$, respectively. Corresponding values from the nonlinear analysis are
 422 $\tau_{zxr} / \sigma_{cf0} = 2.03$ and $\tau_{zxr} / \sigma_{cf0} = 2.21$. Hence, the assumed shear stress in the web at the onset of
 423 yielding in the tension chord ($\tau_{zx} = 4 \text{ MPa}$ (0.58 ksi)) is attained in all cases, though with little margin.

424 Compared to the uniaxially reinforced case, the influence of crack spacing is similar, but somewhat
425 reduced in the nonlinear case (difference on the maximum load about 10%). It can be seen that the
426 deformation capacity is larger, and the increase of shear stresses from cracking to maximum load is
427 substantially more pronounced for the case of where horizontal strains are imposed, with ratios
428 $\tau_{zxi} / \tau_{zxr} \approx (2.03 \dots 2.85) / 0.8 = 2.54 \dots 3.56$, leading to a strain-hardening behavior even in the case of
429 large crack spacings (or if web cracking would occur at much higher loads than assumed here) in
430 spite of $\sigma_{cf0} < f_{ct}$. This indicates that FRC webs without conventional reinforcement should exhibit
431 a much better performance than observed in tests on uniaxially reinforced panels, as long as the chords
432 do not yield; however, this presumption lacks experimental validation as no direct experimental
433 comparison currently exists.

434 Note that substantial compressive longitudinal stresses ($-\Delta\sigma_x$, see Fig. 4c) are obtained due to the
435 longitudinal restraint caused by the imposed strains; these stresses assume the role of the longitudinal
436 reinforcement and must be resisted by the chords, in addition to chord forces induced by bending
437 (similar to the tension shift in girders with transverse reinforcement).

438 COMPARISON WITH EXPERIMENTS

439 In this section, predictions obtained from the simplified compression field analysis for uniaxially
440 reinforced FRC elements outlined above are validated against experimental results of tests carried out
441 by Susetyo et al.^{13,48}. In these studies, eight uniaxially reinforced FRC panels ($\rho_x = 3.31\%$,
442 $\emptyset_x = 8.1 \text{ mm (0.32 in.)}$, $f_s = 552 \text{ MPa (80.1 ksi)}$) and two orthogonally reinforced reference panels
443 measuring $890 \times 890 \times 70 \text{ mm (35} \times 35 \times 3.54 \text{ in.)}$ thick were subjected to in-plane shear²⁹. Two types of
444 concrete (normal and high strength) and three types of hooked-end fibers, with three different fiber
445 dosages were used. The fiber and matrix characteristics of the FRC panels are given in Table 1; it can
446 be seen that most specimens were provided with unpractically high fiber dosages ($\rho_f = 1.5\%$),
447 reflecting the demands of codes mentioned in the introduction.

448 For all predictions, the fiber stresses are strictly determined using Pfyf's model³ with
449 $\tau_{bf} = 2f_{ct} = 0.6f_c^{2/3}$ (in MPa units) and $K_f = 0.375$, without using any modifications or adaptations
450 based on material tests. Furthermore, the predicted crack spacing is determined from Eq. (8). Except
451 for the test specimens with the lowest fiber dosage, the lower bound $s_r = l_f$ in Eq. (8) governs
452 regardless of the considered value of λ . Therefore, in the following analyses, the crack spacing is
453 not varied (i.e. $\lambda = 1$ is considered). For interest, crack spacings observed in the experiments for fiber
454 dosages of 1.00% and 1.50% match fairly well with $s_r = l_f$, as shown in Table 2.

455 Figure 7 compares predictions carried out assuming (i) linear elastic behavior of the concrete in
456 compression as well as (ii) using the parabolic stress-strain relationship, (see Eq. (20)), to describe
457 the concrete in compression with the experimental results of three tests using normal strength concrete
458 with identical fibers, but with variable fiber dosages (i.e. specimens C1F1V1, C1F1V2, C1F1V3).
459 Both models shown in Fig. 7 consider linear elastic reinforcement behavior. While qualitatively the
460 agreement is satisfactory, the maximum loads and deformations are overestimated, particularly when
461 modelling the concrete as linear elastic, where failure is predicted by yielding of the reinforcement.
462 Note that specimen C1F1V1, with $\rho_f = 0.5\%$, remained uncracked during testing until just prior to
463 attaining maximum load, which explains the much stiffer observed behavior compared to the
464 prediction (which only models the post-cracking range). For nonlinear concrete accounting for
465 compression softening, failure is predicted by combined concrete crushing and fiber softening, but
466 deformations are still overestimated (albeit much less than by Reference⁴⁸, in spite of their
467 customization of the tension softening characteristics of the FRC for each individual experiment, also
468 shown in Fig. 7). The most plausible explanation is the premature failure of the elements by sliding
469 along cracks generated in early load-stages, as observed in the tests¹³. Such failures, which are
470 difficult to distinguish from concrete crushing in experiments, are not reflected by rotating-crack
471 models like the CMM-Rf. These models implicitly assume that stresses can be transferred across
472 cracks formed at earlier load stages, whose direction may deviate significantly from the principal

473 direction assumed at higher loads. In order to capture this effect, a model considering fixed,
474 interlocked cracks, such as the CMM-F, would be required. Another explanation could be that with
475 the fiber type used in the three experiments illustrated in Fig. 7, the fiber-fracture condition
476 represented by Eq. (12) is exceeded by roughly 25%, such that the fibers with the longest embedded
477 length would fracture, instead of being pulled out as assumed in the predictions. Although the failure
478 surfaces of the panels were not reported by Susetyo, he noted that the panels did not contain any
479 fractured fibers in any of the dominant cracks¹³. However, as fiber-fracture would only affect a small
480 portion of total amount of fibers bridging the crack it can only explain a small part of the
481 overestimation.

482 Figure 8 compares predictions carried out using the parabolic stress-strain relationship to describe the
483 concrete in compression using Eq. (20), with the experimental results of the remaining tests on FRC
484 panels documented in Reference⁴⁸, all with identical fiber dosage ($\rho_f = 1.5\%$) but different types of
485 hooked end fibers. Again, the agreement is satisfactory, particularly when compared to those in
486 Reference⁴⁸ (shown in Fig. 8), however deformations and maximum loads are overestimated for both
487 the normal strength concrete panels, (refer to Fig. 8a), as well as the high strength concrete specimens
488 shown in Fig. 8b. Possible explanations for the normal strength concrete panels are the same as
489 discussed above. The more pronounced differences in the high strength concrete specimens could be
490 explained by the smoother crack faces in high strength concrete, caused by cracks passing through
491 the aggregates (rather than around them as is the case for normal strength concrete)⁴⁹. Furthermore,
492 in Specimen C2F1V3-R, the fiber-fracture condition given by Eq. (12) is violated by roughly 70%,
493 such that the majority of the fibers in this case would fracture rather than pull out as assumed in the
494 predictions; this could basically explain the overestimation of maximum load as well as deformations.
495 However, similar results are obtained for Specimen C2F2V3, where only about 75% of the tensile
496 strength of the fibers is reached according to Eq. (12). Hence, this explanation does not appear to be
497 conclusive.

498 For interest, the predicted responses by Lee et al.⁵⁰ using a finite element approach are also provided
499 in Figs. 7 and 8. In their model, Lee et al. considered crack slip, which essentially represents a fixed
500 crack model, however, disregarded the interaction between fiber stresses and aggregate interlock in
501 the transfer of shear across cracks. Despite the excellent correlation with the test data, it is
502 questionable whether the semi-empirical approach by Lee is extendable to other, more practical
503 dosages of SFRC.

504 **SUMMARY AND CONCLUSIONS**

505 This paper presents a simplified rotating crack model for the analysis and design of steel fiber
506 reinforced elements in shear (CMM-Rf) based on the Cracked Membrane Model with stress-free
507 cracks (CMM-R)^{23,24}. Simplified solution procedures of the CMM-Rf are developed for (i) uniaxially
508 reinforced FRC elements in pure shear and (ii) FRC elements without conventional reinforcement,
509 but with imposed longitudinal strains. It is shown that the applied load can be increased after cracking
510 even when using practical, moderate fiber dosages. This is due to the fact that more fibers can be
511 activated by progressively rotating the principal compressive stress direction towards the direction of
512 the uniaxial reinforcement or imposed strain, respectively. A hardening response is thus obtained
513 until the point where due to the softening of the fibers, excessively flat inclinations of the principal
514 compressive stress direction are required to maintain equilibrium, resulting in failures by concrete
515 crushing particularly if compression softening is accounted for.

516 The model is compared with available test data and shows a satisfactory correlation. However, it is
517 revealed that rotating crack models are not capable of capturing failures which are governed by sliding
518 of cracks in elements with high fiber dosages. In order to cover such cases, it is proposed to extend
519 the general Cracked Membrane Model (CMM-F)^{23,24}, which considers interlocked cracks with a fixed
520 orientation, to account for fiber reinforcement, in a future study.

521 Finally, more experimental data is required in order to validate theoretical models, particularly with
522 practical and economical fiber dosages, as well as to investigate the behavior of FRC elements
523 without any conventional reinforcement, but with imposed longitudinal strains.

524

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- 653

654 **APPENDIX**

655 This Appendix contains additional information that were not provided in the main body of the paper
656 for the sake of brevity.

657 **Notation**

658	d_f	=	diameter of fiber
659	E_c	=	elastic modulus of concrete
660	E_f	=	elastic modulus of fibers
661	E_s	=	elastic modulus of steel reinforcement
662	E_{sh}	=	hardening modulus of steel reinforcement
663	K_f	=	fiber orientation factor
664	f_c	=	compressive strength of concrete matrix
665	f_{ct}	=	tensile strength of concrete matrix
666	f_s	=	yield strength of longitudinal reinforcement
667	f_t	=	ultimate strength of longitudinal reinforcement
668	f_{tf}	=	ultimate strength of fiber reinforcement
669	l_f	=	length of fiber
670	l_{fb}	=	embedded length of fiber
671	n	=	modular ratio
672	s_r	=	diagonal crack spacing without fibres
673	s_{r0}	=	maximum diagonal crack spacing without fibres
674	s_{rf}	=	diagonal crack spacing for SFRC members
675	s_{rf0}	=	maximum diagonal crack spacing for SFRC members
676	u	=	orthogonal crack opening
677	V_R	=	total shear force
678	V_{Rc}	=	ultimate shear force carried by concrete
679	V_{Rf}	=	ultimate shear force carried by fibers
680	V_{Rs}	=	ultimate shear force carried by transverse reinforcement
681	w_r	=	crack opening
682	ε	=	strain
683	ε_{c0}	=	strain of concrete corresponding to the peak compressive stress
684	ε_{su}	=	ultimate tensile strain of longitudinal reinforcement

685	γ_{zx}	=	shear strain
686	λ	=	crack spacing coefficient
687	σ_c	=	stress in concrete
688	σ_{cf}	=	effective (residual) fiber stresses at the cracks
689	σ_s	=	stress in steel reinforcement
690	ρ_f	=	volumetric fiber dosage
691	$\rho_{f,\min}$	=	minimum volumetric fiber dosage for SFRC members
692	ρ_{\min}	=	minimum longitudinal reinforcing ratio
693	ρ_x	=	longitudinal reinforcing ratio in x -direction
694	ρ_z	=	longitudinal reinforcing ratio in z -direction
695	τ_b	=	bond shear stress for reinforcing bar and matrix
696	τ_{bf}	=	bond shear stress for fiber and matrix
697	τ_{zx}	=	applied shear stress
698	θ	=	principal compressive stress and strain direction
699	θ_f	=	fiber inclination
700	θ_r	=	crack direction
701	\emptyset	=	steel reinforcing bar diameter

702

703 **Calculation of fiber orientation factor**

704 Assuming that all possible fiber orientations have an equal probability of occurrence (either in 2D or
705 3D, or, in few cases, in-between), the fiber orientation factor, K_f , is obtained by integrating all
706 possible fiber orientations, observing that the amount of fibers with an inclination θ_f crossing a crack
707 plane of unit surface amounts to $\cos\theta_f$, see Fig. A.1a. Discounting the fibers whose inclination
708 exceeds $|\theta_f| > \theta_{eff}$, one obtains

$$709 \quad K_f = \frac{1}{\pi} \int_{-\theta_{eff}}^{\theta_{eff}} \cos\theta_f d\theta_f = \frac{2}{\pi} \sin\theta_{eff} \left(\theta_{eff} = \frac{\pi}{2} : K_f = \frac{2}{\pi}; \quad \theta_{eff} = \frac{\pi}{3} : K_f = \frac{\sqrt{3}}{\pi} \right) \quad (A.1)$$

710 for a 2D fiber distribution and

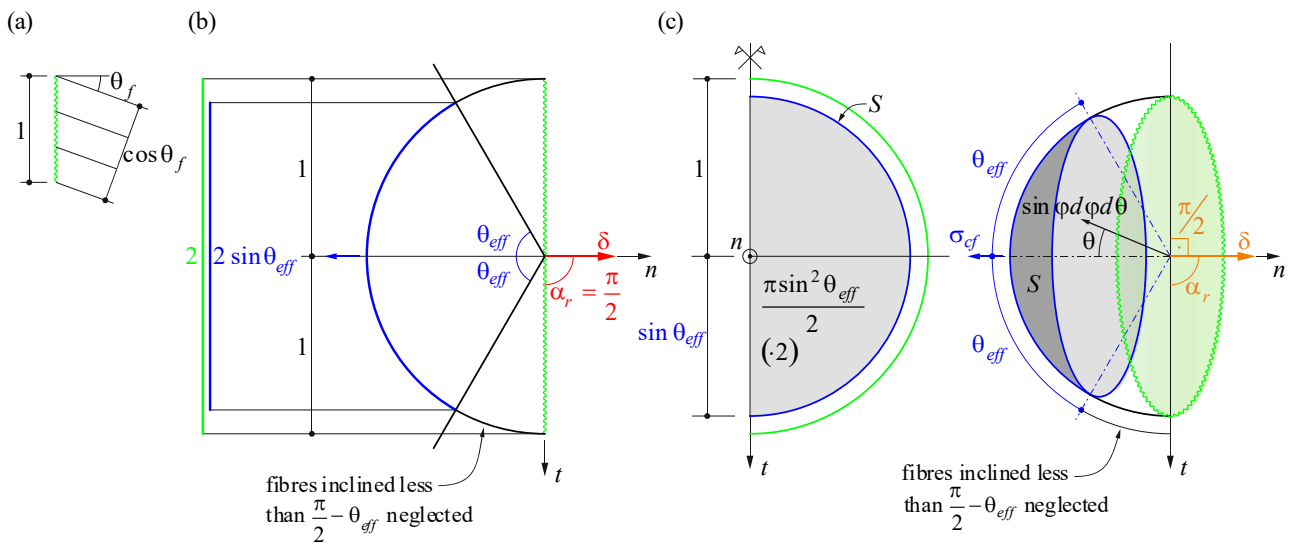
711
$$K_f = \frac{1}{2\pi} \int_{-\pi}^{\pi} \int_0^{\theta_{eff}} \cos \theta_f dA = \frac{1}{2} \sin^2 \theta_{eff} \left(\theta_{eff} = \frac{\pi}{2} : K_f = \frac{1}{2}; \quad \theta_{eff} = \frac{\pi}{3} : K_f = \frac{3}{8} \right) \quad (A.2)$$

712 for a 3D fiber distribution, where $dA = \sin \theta_f \cdot d\theta_f \cdot d\varphi$ in spherical coordinates has been used.

713 Note that Eqs. (A.1) and (A.2) can be interpreted as the projection of the surface S defined by the

714 fiber end loci of all effective fibers to the crack plane (a semi-circle and a semi-sphere for the 2D and

715 3D case), divided by the surface S , see Fig. A.1b and c, respectively.



716

717 *Fig. A.1–Fiber orientation factor for orthogonal crack opening: (a) notation, (b) 2D and (c) 3D fiber*

718 *distribution, respectively, neglecting fibers oriented close to the crack plane.*

719

TABLES AND FIGURES

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744 Fig. 6 – FRC element without conventional reinforcement, but imposed horizontal strains,
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 758 problems in initial test).

Specimen	f_c [MPa] ([ksi])	$-\epsilon_{c0}$ ($\times 10^{-3}$)	ρ_f (%)	d_f [mm] ([in.])	l_f [mm] ([in.])	f_{tf} [MPa] ([ksi])
C1F1V1	51.4 (7.45)	2.15	0.50	0.62 (0.024)	50 (1.97)	1050 (152.3)
C1F1V2	53.4 (7.74)	2.67	1.00	0.62 (0.024)	50 (1.97)	1050 (152.3)
C1F1V3	49.7 (7.21)	2.50	1.50	0.62 (0.024)	50 (1.97)	1050 (152.3)
C1F2V3	59.7 (8.66)	3.27	1.50	0.38 (0.015)	30 (1.18)	2300 (333.6)
C1F3V3	45.5 (6.60)	2.34	1.50	0.55 (0.022)	35 (1.38)	1100 (159.5)
C2F1V3-R	78.8 (11.43)	2.78	1.50	0.62 (0.024)	50 (1.97)	1050 (152.3)
C2F2V3	76.5 (11.10)	2.22	1.50	0.38 (0.015)	30 (1.18)	2300 (333.6)
C2F3V3	62.0 (8.99)	2.03	1.50	0.55 (0.022)	35 (1.38)	1100 (159.5)

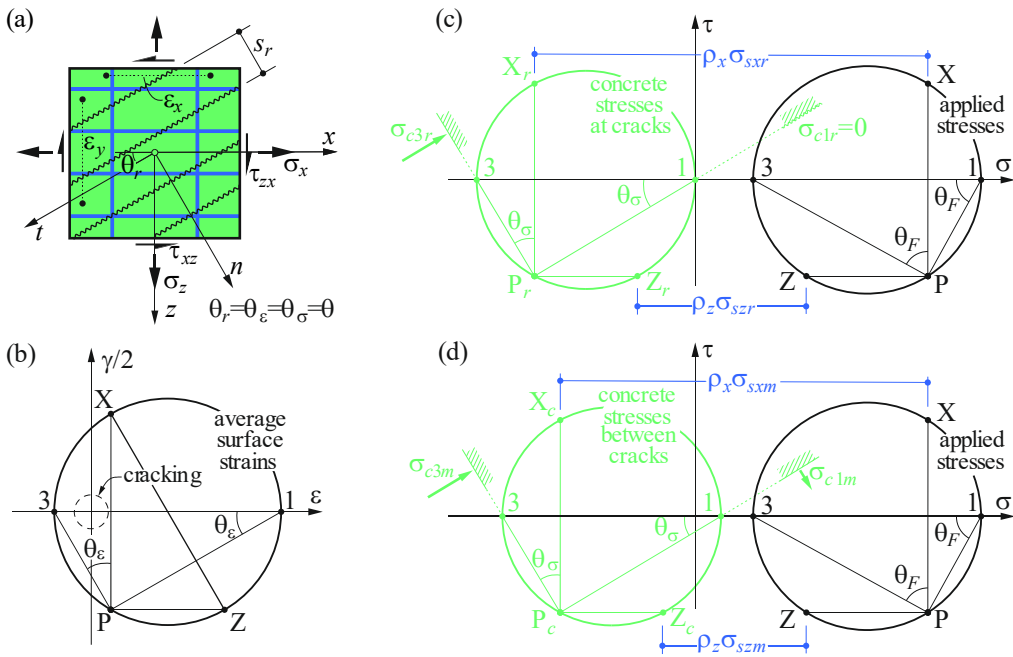
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760 Table 2 – Comparison of predicted crack spacing, s_{rf} , and observed average crack spacing, $s_{r,exp}$, for
 761 tests by Susetyo et al.⁴⁸.

Specimen	$s_{r,exp}$ [mm] ([in.])	$s_{rf}(\lambda=0.5)$ - Eq. (8) [mm] ([in.])	$s_{rf}(\lambda=1.0)$ - Eq. (8) [mm] ([in.])
C1F1V1	114 (4.49)	146 (5.75)	73 (2.87)
C1F1V2	55 (2.17)	50 (1.97)	50 (1.97)
C1F1V3	57 (2.24)	50 (1.97)	50 (1.97)
C1F2V3	38 (1.50)	30 (1.18)	30 (1.18)
C1F3V3	57 (2.24)	35 (1.38)	35 (1.38)
C2F1V3-R	36 (1.42)	50 (1.97)	50 (1.97)
C2F2V3	47 (1.85)	30 (1.18)	30 (1.18)
C2F3V3	41 (1.61)	35 (1.38)	35 (1.38)

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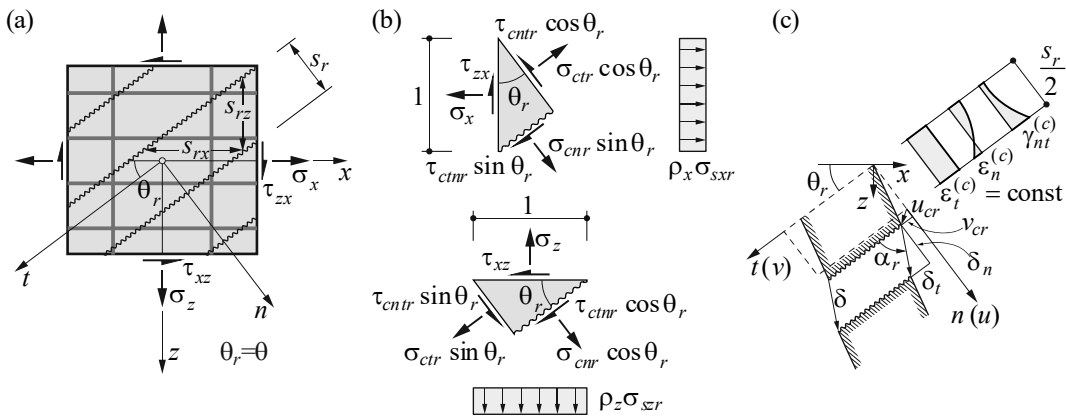
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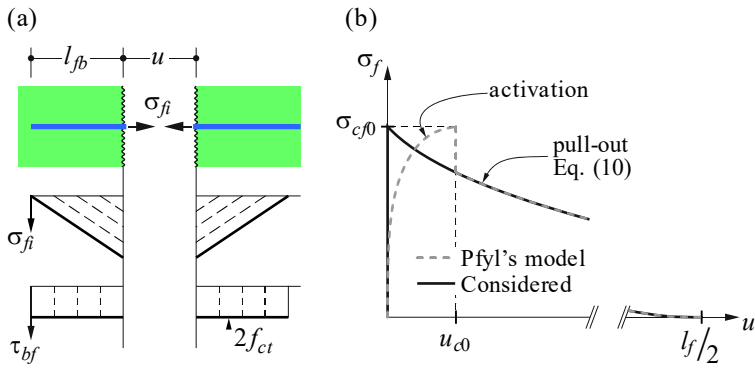
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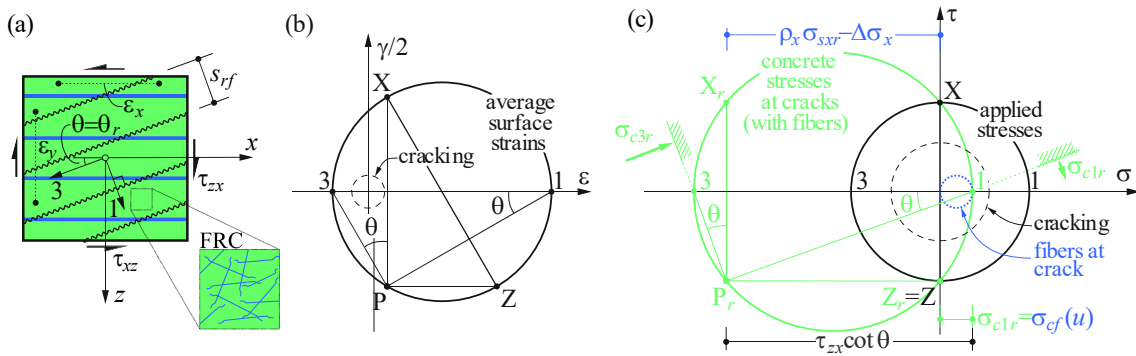
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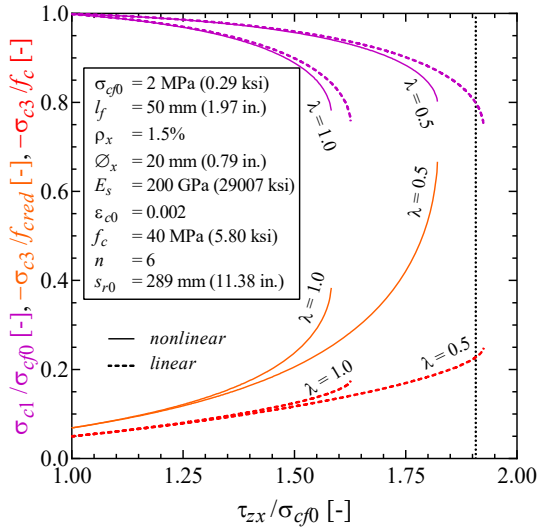
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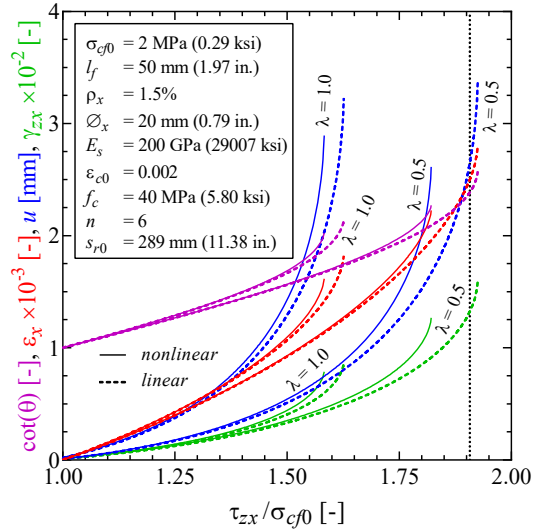
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779 surface strains; (c) stresses at cracks.

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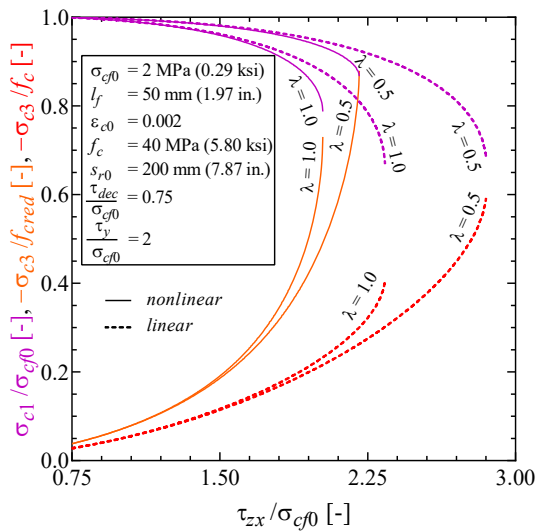


(a)

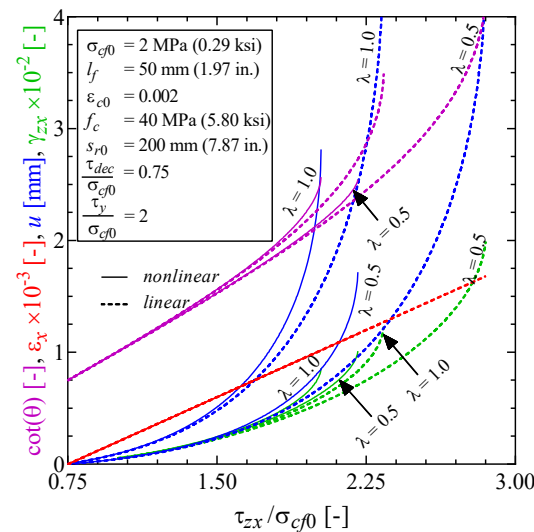


(b)

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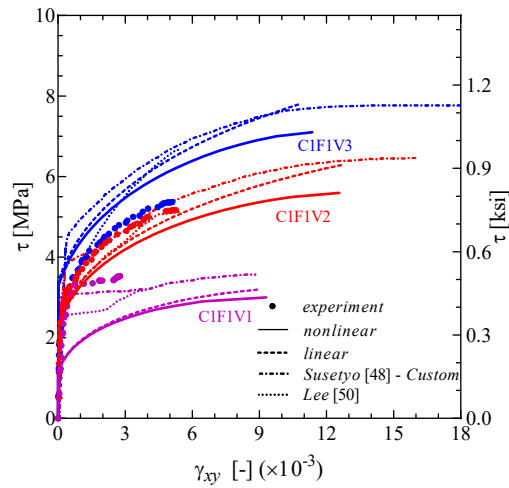


(a)



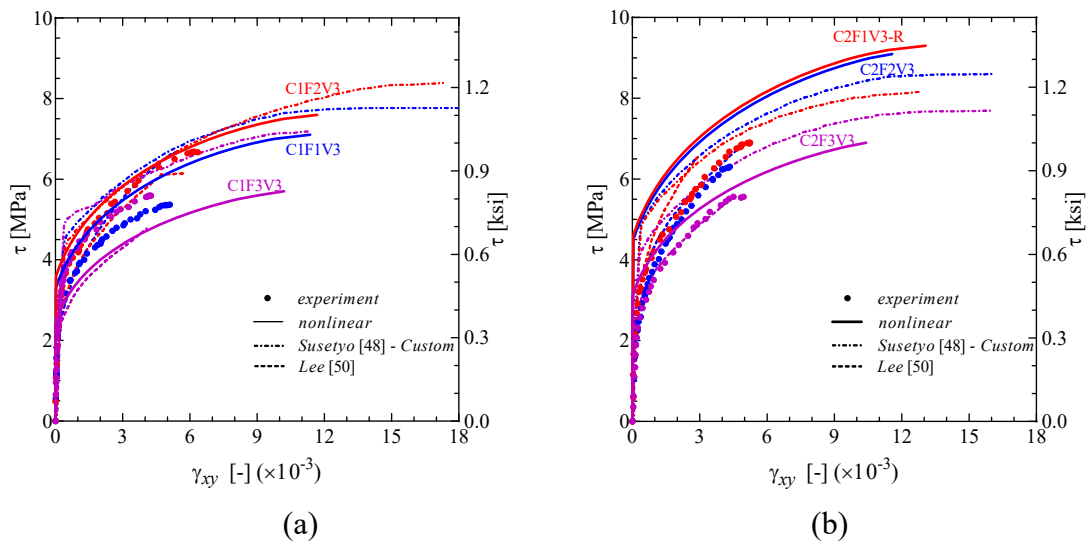
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